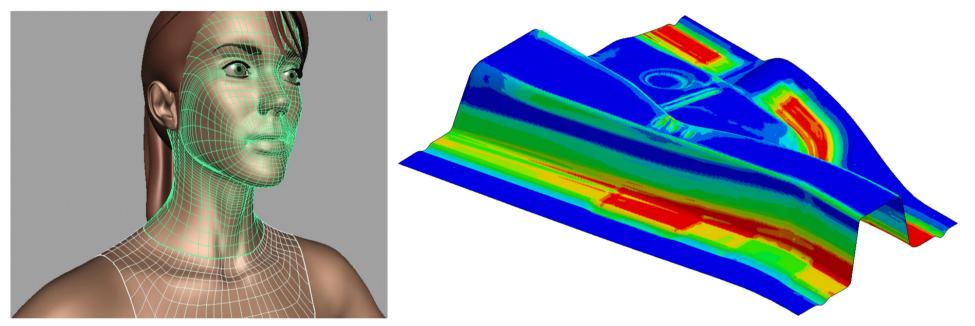
# Recent developments in LS-DYNA for Isogeometric Analysis



T.J.R. Hughes

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Some slides borrowed from:

T.J.R. Hughes:Professor of Aerospace Engineering and Engineering Mechanics, University of Texas at AustinD.J. Benson:Professor of Applied Mechanics, University of California, San Diego

Recent developments in LS-DYNA for Isogeometric Analysis



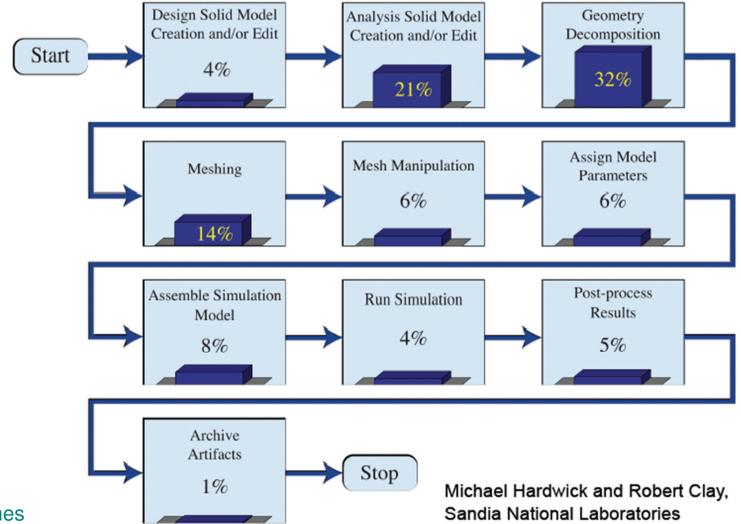
# Outline

- Isogeometric Analysis
  - motivation / definition / history
- From B-splines to NURBS (T.J.R. Hughes)
   basis functions / control net / refinements
- generalized elements in LS-DYNA (D.J. Benson)
   basic idea / shell formulations / interpolation-nodes / interpolation-elements
- NURBS-based finite elements in LS-DYNA
  \*ELEMENT\_NURBS\_PATCH\_2D
- Example: Underbody Cross Member (Numisheet 2005)
   description / comparison of results / summary
- Summary and Outlook



#### Isogeometric Analysis – motivation (... at the beginning)

reduce effort of geometry conversion from CAD into a suitable mesh for FEA



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#### Isogeometric Analysis - definition

#### ISOPARAMETRIC (FE-Analysis)

use same approximation for geometry and deformation

(normally: low order Lagrange polynomials ---- in LS-DYNA basically only linear elements)

GEOMETRY  $\leftarrow \rightarrow$  DEFORMATION

#### ISOGEOMETRIC (CAD - FEA)

same description of the geometry in the design (CAD) and the analysis (FEA)

CAD  $\leftrightarrow$  FEA

#### common geometry descriptions in CAD

- NURBS (Non-Uniform Rational B-splines) → most commonly used
- T-splines

→ enhancement of NURBS

- subdivision surfaces
- and others

 $\rightarrow$  mainly used in animation industry



## Isogeometric Analysis - history

- start in 2003
  - summer: Austin Cotrell starts as PhD Student of Prof. T.J.R. Hughes at the University of Texas, Austin
  - autumn: first NURBS-based FE-code for linear, static problems provides promising results, the name "ISOGEOMETRIC" is used the first time
- 2004 up to now: many research activities to various topics
  - non-linear structural mechanics
    - ightarrow shells with and without rotational DOFs
    - $\rightarrow$  implicit gradient enhanced damage
    - $\rightarrow$  XFEM
  - shape- und topology-optimization
  - efficient numerical integration
  - turbulence and fluid-structure-interaction
  - acoustics
  - refinement strategies
  - ...
- January 2011: first thematic conference on *Isogeometric Analysis* 
  - "Isogeometric Analysis 2011: Integrating Design and Analysis", University of Texas at Austin



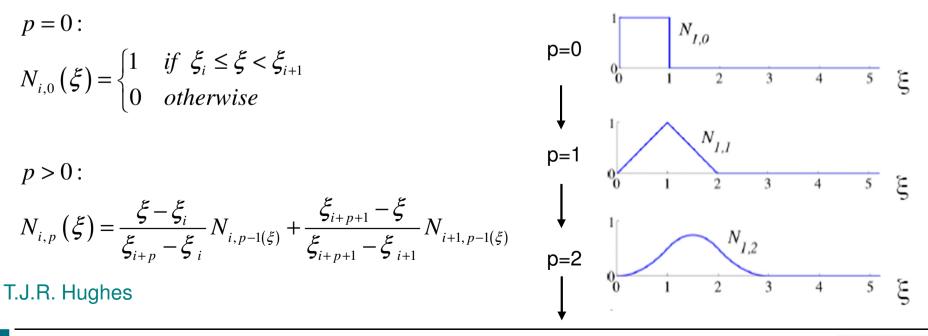
#### B-spline basis functions

- constructed recursively
- positive everywhere (in contrast to Lagrange polynomials)
- shape of basis funktions depend on: knot-vector and polynomial degree
- knot-vector: non-decreasing set of coordinates in parameter space
- normally C<sup>(P-1)</sup>-continuity
  - $\rightarrow$  e.g. lin. / quad. / cub. / quart. Lagrange:
  - $\rightarrow$  e.g. lin. / quad. / cub. / quart. B-spline:

 $\rightarrow C^0 / C^0 / C^0 / C^0$  $\rightarrow C^0 / C^1 / C^2 / C^3$ 

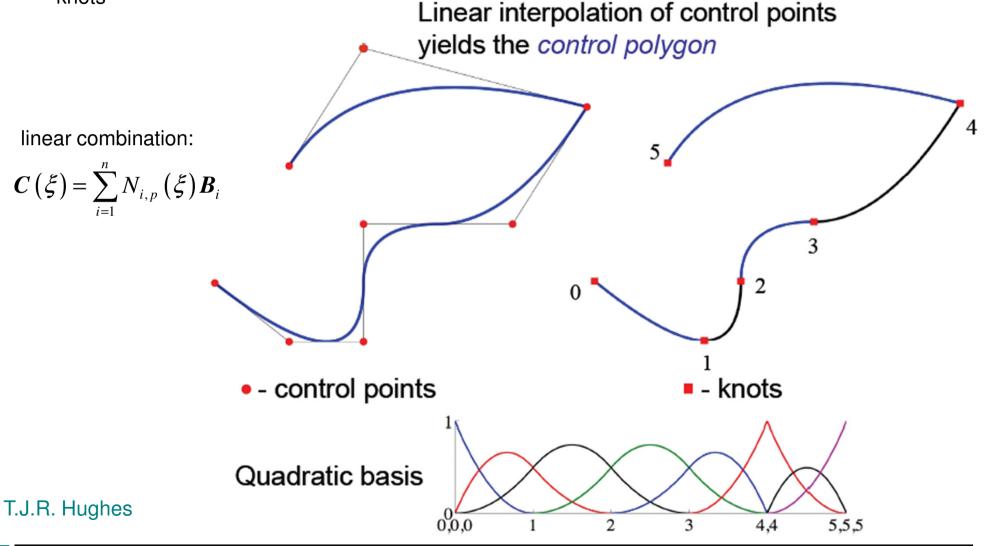
Example of a uniform knot-vector:

$$\Xi = \{0, 1, 2, 3, 4, \dots\}$$





- B-spline curves
  - control points  $\boldsymbol{B}_i$  / control polygon (control net)
  - knots



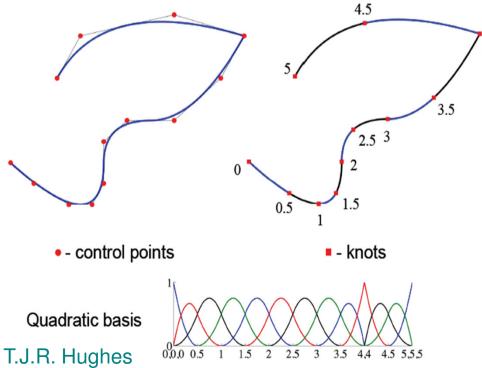


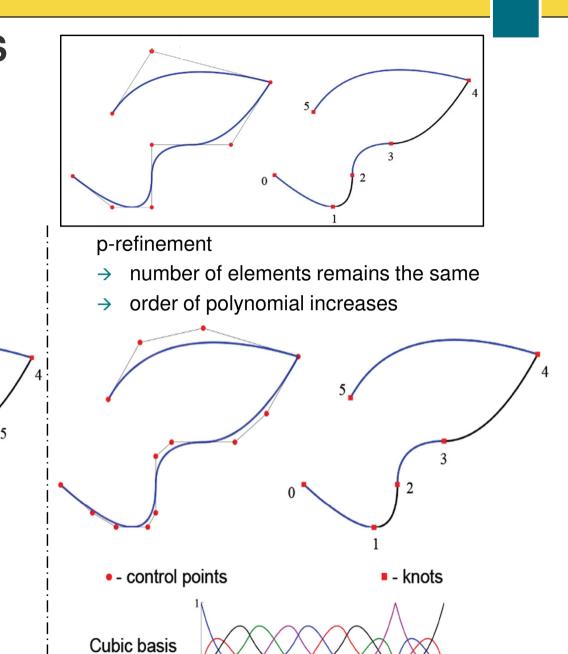
B-splines

- refinement posibilities

h-refinement

- → number of elements increases
- → order of polynomial remains the same





1.1

2.2

3.3

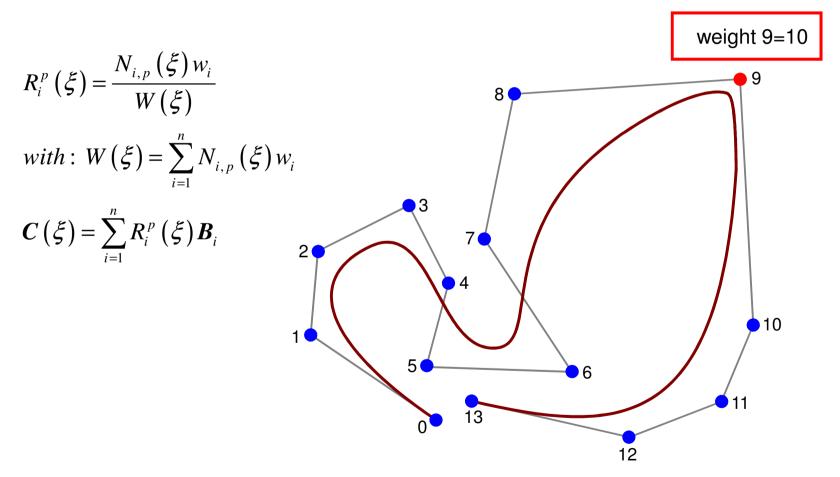
4,4,4

0.0.0.0



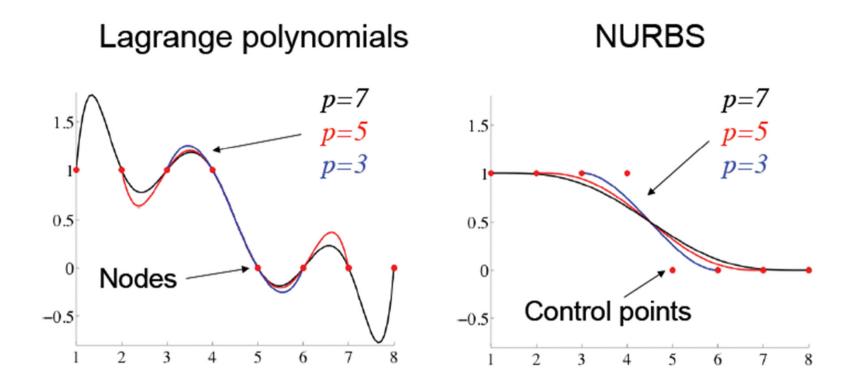
5.5.5.5

- NURBS Non-Uniform Rational B-splines
  - weights at control points leads to more control over the shape of a curve
  - projective transformation of a B-spline





smoothness of Lagrange polynomials vs. NURBS

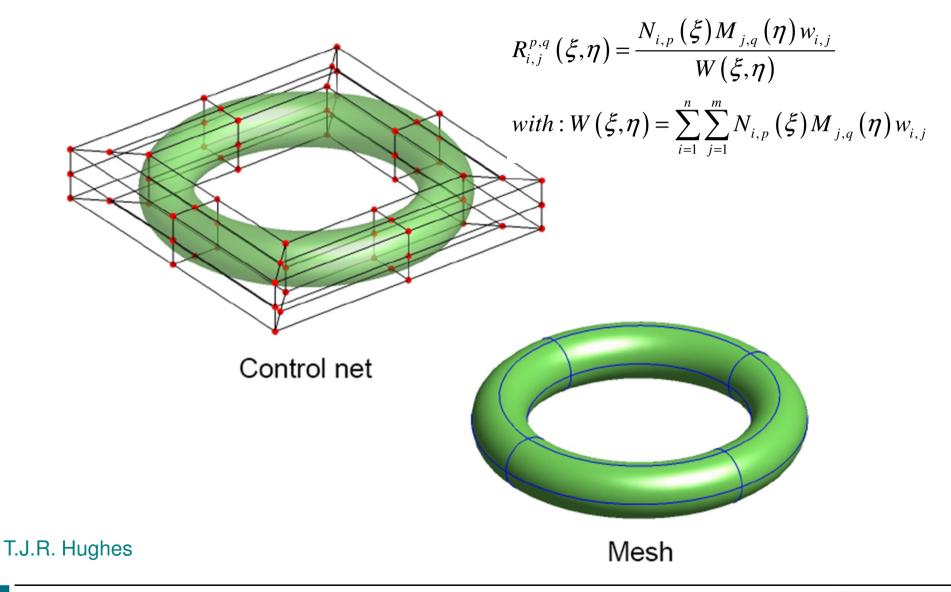


#### T.J.R. Hughes



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NURBS – surfaces (tensor-product of univariate basis)



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# From B-splines to NURBS - summary

#### B-spline basis functions

- recursive
- dependent on knot-vector an polynomial order
- normally  $C^{(P-1)}$ -continuity
- "partition of unity" (like Lagrange polynomials)
- refinement (h/p and k) without changing the initial geometry  $\rightarrow$  adaptivity
- control points are normally not a part of the physical geometry (non-interpolatory basis functions)

#### NURBS

- B-spline basis functions + control net with weights
- all mentioned properties for B-splines apply for NURBS



# generalized elements in LS-DYNA

- basis functions
  - active research as well in the field of CAD as in computer animation
- implementation of finite elements for specific basis functions
  - time consuming
  - software might become obsolete once new types of basis functions appear
  - wish: possibility for fast prototyping of new elements

# $\int$

- generalized elements (shells and solids)
  - elements are formulated with the help of generalized coordinates
  - implementation is independent of utilized basis functions
  - new basis functions can be used immediately  $\rightarrow$  rapid prototyping of elements
  - properties of generalized elements are defined exclusively in the input deck:
  - $\rightarrow$  values of shape functions and its derivatives at generalized coordinates and integration points
  - $\rightarrow$  values of integration weights
  - generalized formulation allows the use of different types of basis functions
    - $\rightarrow$  Lagrange polynomials (standard FEA) / NURBS / T-splines / subdivision surfaces / ...

#### D.J. Benson



# generalized elements in LS-DYNA - visualization

generalized coordinates (coorespond to control points in NURBS)
 - are normally not part of the physical geometry

#### LS-PrePost

- displays only elements with linear basis functions
- $\rightarrow$  right now it is able to display and modify NURBS
- interpolation elements
  - linear elements to visualize results of generalized elements
  - are used for contact treatment

#### interpolation nodes

- nodes on physical geometry to define interpolation elements
- displacements of interpolation nodes follow a linear function depending on displacements of generalized coordinates



# generalized elements in LS-DYNA

#### \*DEFINE\_ELEMENT\_GENERALIZED\_SHELL

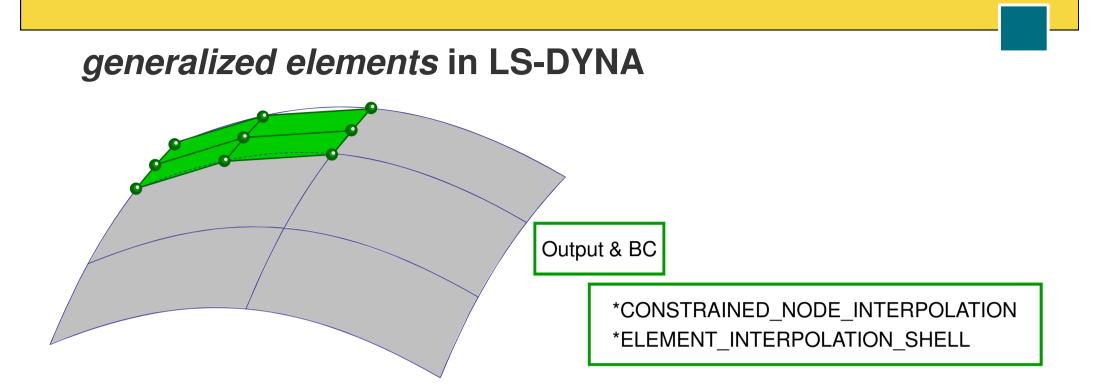
- element ID / number of IPs / number of generalized coordinates
- for each IP: weights & values of all basis functions and derivatives at IPs
- for each generalized coordinate: values of all basis functions and derivatives at control points

 $\rightarrow$  e.g.: a typical 9-noded element with 9 IPs necessitates approx. **172 lines** in input deck!

#### \*ELEMENT\_GENERALIZED\_SHELL

- connectivity of an element (here: blue control points)





- \*CONSTRAINED\_NODE\_INTERPOLATION
  - for each interpolation node: number of control points of which the position of this interpolation node is dependent
  - IDs of control points and weighting factors
    - → displacement of interpolation node will be interpolated linearly depending on the displacements of control points
  - will be used for contact treatment at the moment
- \*ELEMENT\_INTERPOLATION\_SHELL
  - dependency to control points with appropriate weighting factors



# generalized elements in LS-DYNA – shell formulations

- shear deformable & thin shell theory
  - with rotational DOFs
  - without rotational DOFs

	shear deformable shell theory	thin shell theory	
displacement field: $x(\eta, \xi, \zeta)$	$\sum_{i} N_{i}(\eta,\xi) \left( x_{i} + \frac{h}{2} \zeta \hat{y}_{i} \right)$	$\sum_{i} N_{i}(\eta,\xi) x_{i} + \frac{h}{2} \zeta \hat{n}(\eta,\xi)$	
velocity field: $\dot{x}(\eta, \xi, \zeta)$	$\sum_{i} N_{i}(\eta,\xi) \left( \dot{x}_{i} + \frac{h}{2} \zeta \dot{\hat{y}}_{i} \right)$	$\sum_{i} N_{i}(\eta,\xi) \dot{x}_{i} + \frac{h}{2} \zeta \dot{\hat{n}}(\eta,\xi)$	
unit orientation vector (shell normal):	ŷ	ĥ	
	with /without rotational DOFs	with / without rotational DOFs	
time derivative of unit orientation vector:	$\dot{\hat{y}}_i = \omega_i \times \hat{y}_i$ $\dot{\hat{y}}_i = \sum_j \frac{\partial \hat{y}_i}{\partial x_j} \dot{x}_j$		



# generalized elements in LS-DYNA

#### analysis capabilities

- implicite and explicit time integration
- eigenvalue analysis
- many material models from the LS-DYNA material library are available
- some boundary conditions are implemented via *interpolation elements* 
  - contact treatment
  - pressure distribution (not fully tested yet)
- time step control via "maximum system eigenvalue"
  - D.J. Benson: Stable Time Step Estimation for Multi-material Eulerian Hydrocodes, CMAME,191-205 (1998)

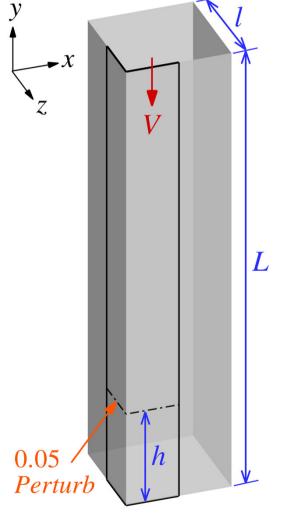
#### generalized *solids* are implemented as well

- "standard" displacement elements



# generalized elements in LS-DYNA - Example

buckling of a square tube (using NURBS basis functions)



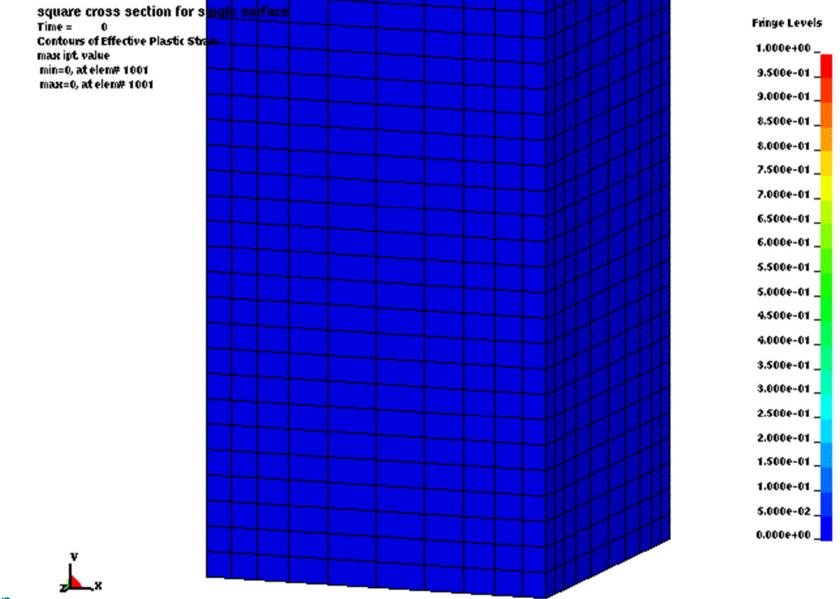
#### standard benchmark for automobile crashworthiness

- quarter symmetry to reduce cost
- perturbation to initiate buckling mode
- J<sub>2</sub> plasticity with linear isotropic hardening
- mesh:
  - 640 quartic (P=4) elements.
  - 1156 control points.
  - 3 integration points through thickness.

#### D.J. Benson

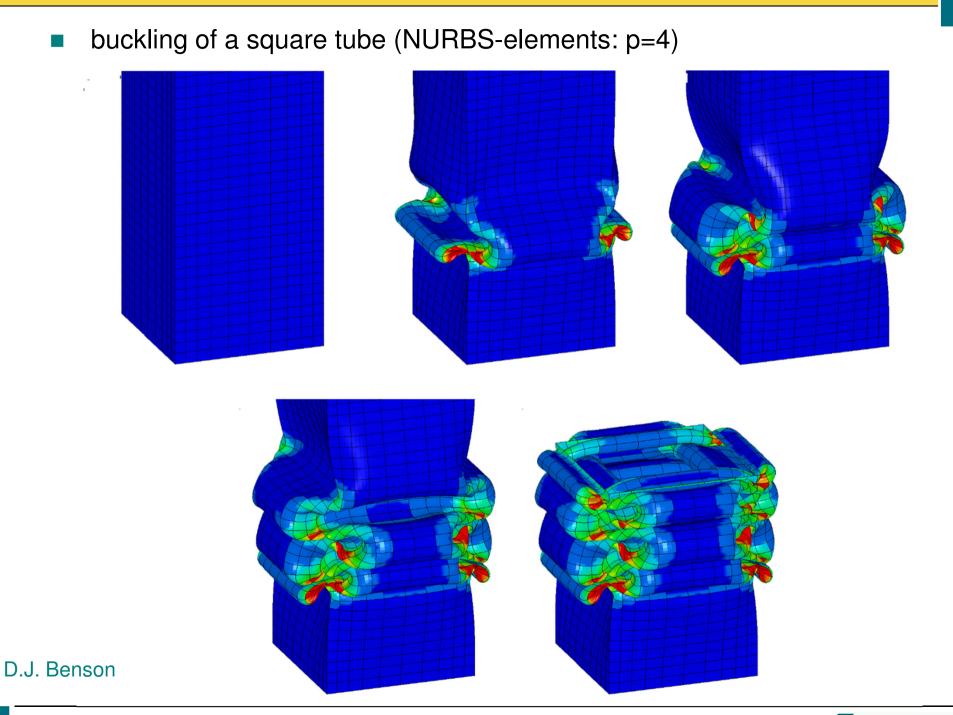


#### buckling of a square tube (NURBS-elements: p=4)



D.J. Benson





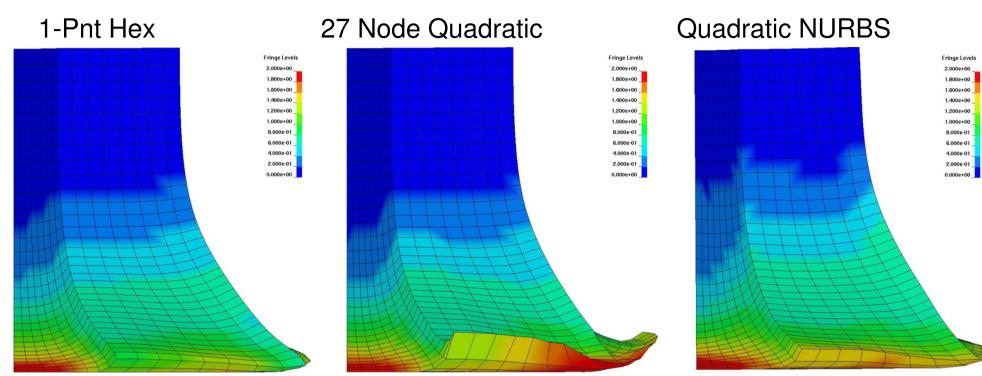
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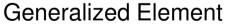
# generalized elements in LS-DYNA - solid elements

Taylor Bar Impact



**Generalized Element** 

Standard LS-DYNA element



Formulation	# Nodes/ CP	Peak Plastic Strain	# Time Steps	
1-Pnt Hex	2677	2.164	2136	
Quad. Lagr.	2677	2.346	3370	
Quad. NURBS	648	2.479	954	

D.J. Benson



# generalized elements in LS-DYNA - summary

#### fast prototyping of new elements

- all the information in the input deck
- no limitation on number of control points and integration points per element
- no restriction to special types of basis functions
- interesting for research
- rather difficult to create input deck  $\rightarrow$  not usable for industry
- good results with NURBS  $\rightarrow$  decision to implement NURBS-based finite elements in LS-DYNA

#### generalized shells

- shear deformable and thin shell theories implemented
- with and without rotational DOFs

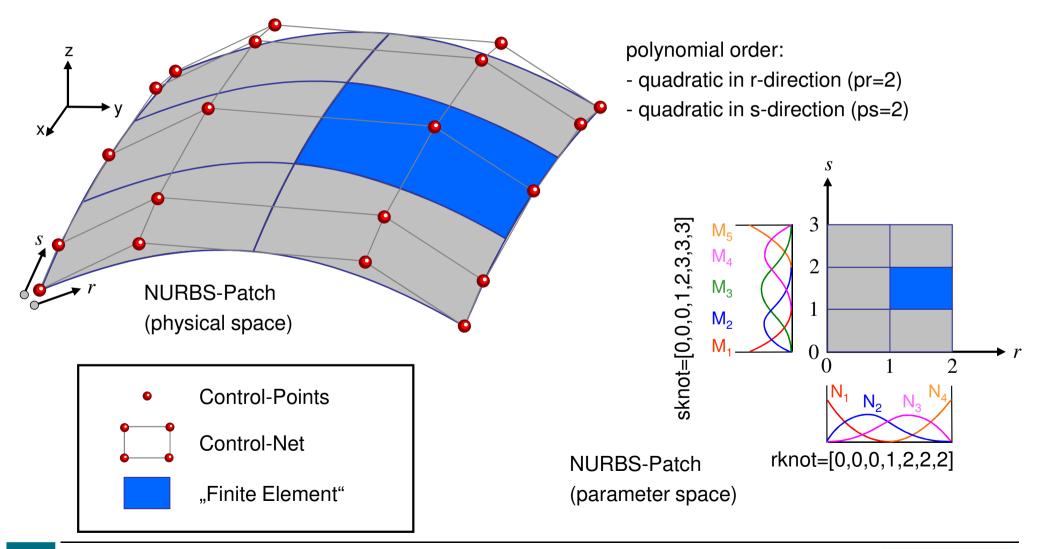
#### generalized solids

- "standard" displacement formulation

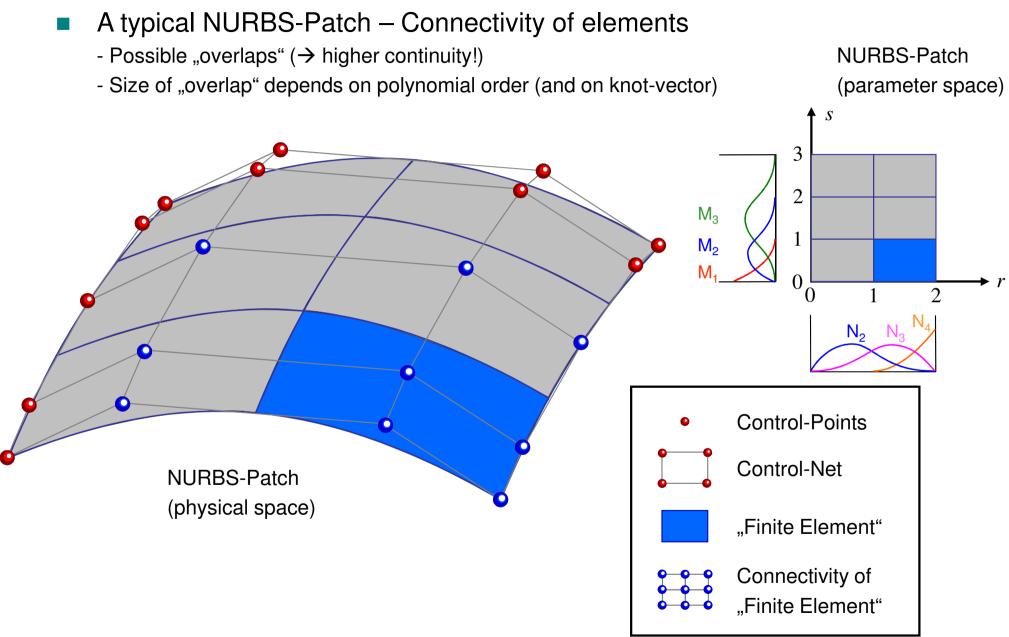




- A typical NURBS-Patch and the definition of elements
  - elements are defined through the knot-vectors (interval between different values)
  - shape functions for each control-point









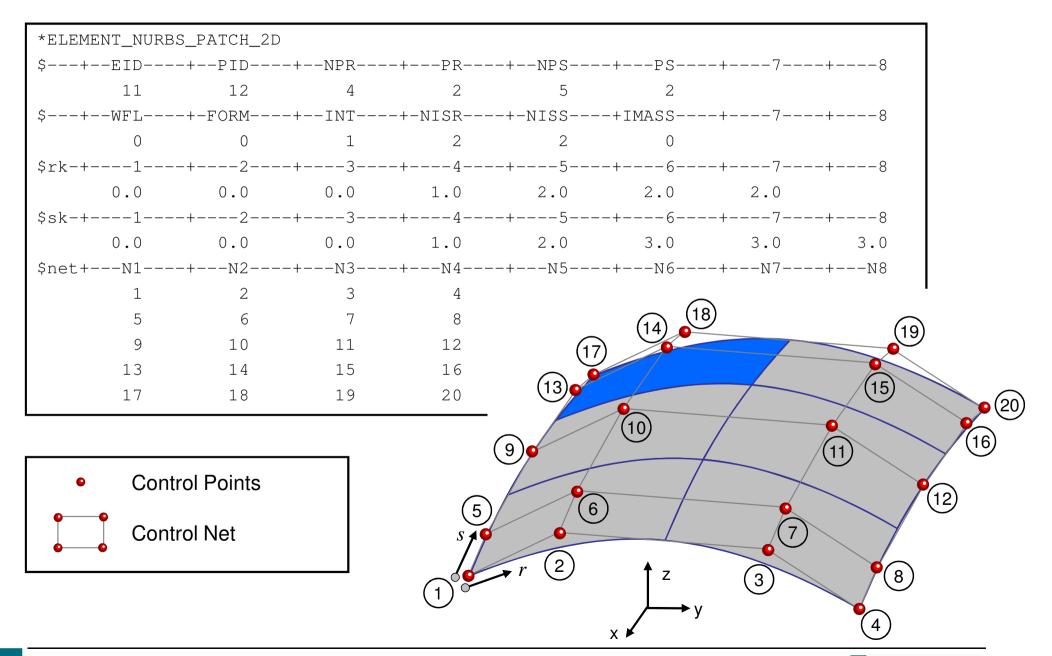
#### New Keyword: \*ELEMENT\_NURBS\_PATCH\_2D

- definition of NURBS-surfaces
- 4 different shell formulations with/without rotational DOFs ( $\rightarrow$  generalized shells)

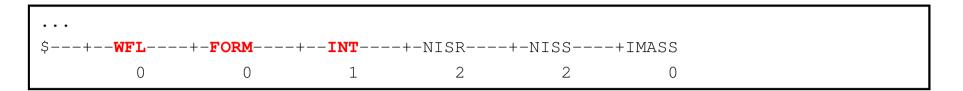
#### Pre- and Postprocessing

- work in progress for LS-PrePost ... current status (Ispp3.1beta)
  - $\rightarrow$  visualization of 2D-NURBS-Patches
  - → import IGES-format and construct \*ELEMENT\_NURBS\_PATCH\_2D
  - → modification of 2D-NURBS geometry
  - $\rightarrow$  ... much more to come!
- Postprocessing and boundary conditions (i.e. contact) currently with
  - interpolation nodes
  - interpolation elements
- Analysis capabilities (→ generalized shells)
  - implicit and explicit time integration
  - eigenvalue analysis
  - other capabilities (e.g. geometric stiffness for buckling) implemented but not yet tested
- LS-DYNA material library available (including umats)









- WFL Flag for weighting factors for control points
  - 0: All weights are 1.0 (no need to define them  $\rightarrow$  B-splines)
  - 1: define weights for control points

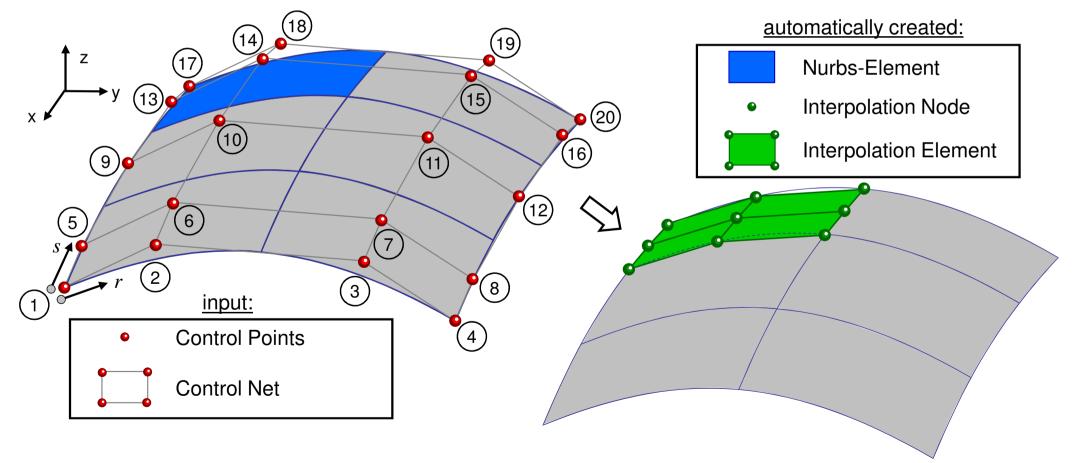
#### FORM – Shell formulation to be used

- 0: "shear deformable theory" with rotational DOFs
- 1: "shear deformable theory" without rotational DOFs
- 2: "thin shell theory" without rotational DOFs
- 3: "thin shell theory" with rotational DOFs
- INT In-plane integration rule
  - 0: reduced (Gauss-)integration (NIP=PR\*PS)
  - 1: full (Gauss-)integration (NIP=(PR+1)\*(PS+1)
  - -?: "Half-Point-Rule" ( $\rightarrow$  A. Reali)

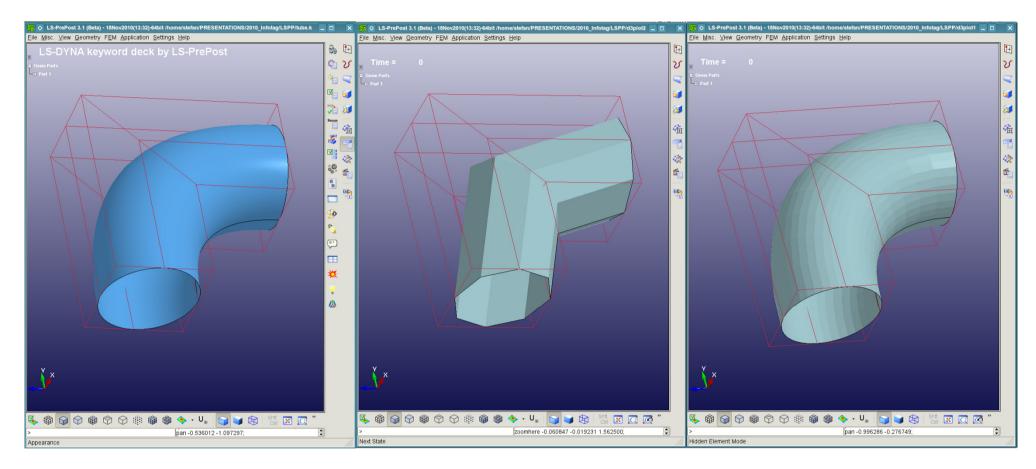




 NISR/NISS – Number of Interpolation Elements per Nurbs-Element (r-/s-dir.) important for post-processing, boundary conditions and contact treatment







#### LSPP: Preprocessing

- control-net
- nurbs surface

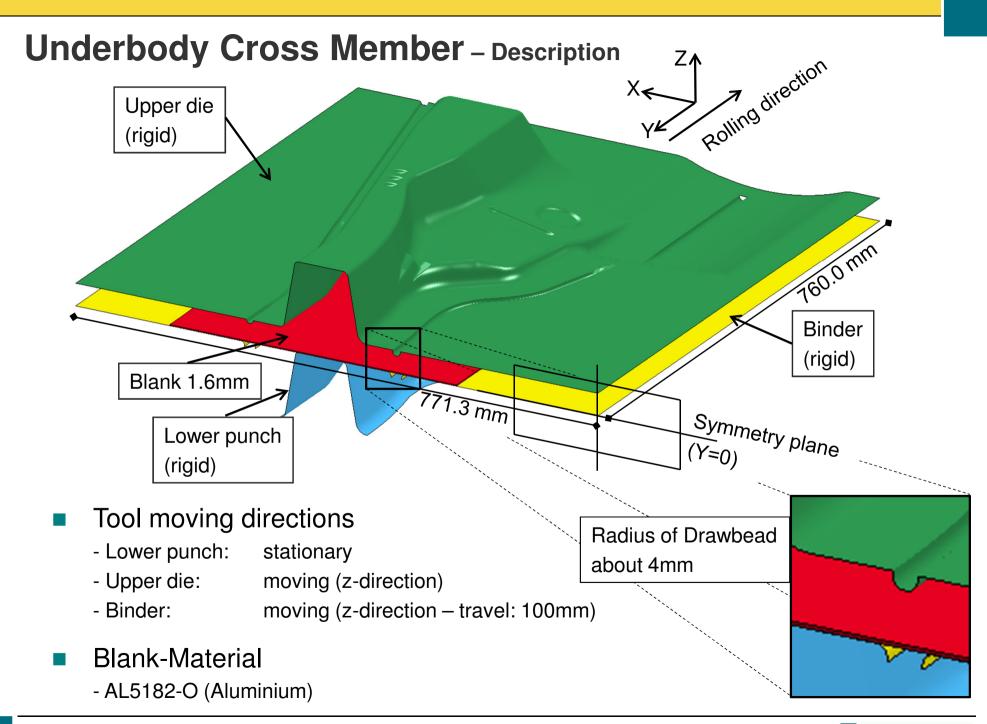
#### LSPP: Postprocessing

Interpolation nodes/elements

nisr=niss=2

nisr=niss=10







#### **Underbody Cross Member** – Simulation models

#### identical for all

- material model: \*MAT\_TRANSVERSELY\_ANISOTROPIC\_ELASTIC\_PLASTIC (\*MAT\_037)
- nip=5 number of integration points through the thickness
- istupd=0 no thickness update
- imscl=0 no "selective" mass scaling (no mass scaling at all!)
- SMP, double precision, ncpu=4 (Dual Core AMD Opteron, 2.2 GHz)

#### standard elements

- ELFORM=16: fully integrated (4-noded) shell-elements with assumed strain formulation
- discretizations: with adaptivity (mesh size:  $4mm \rightarrow 2mm \rightarrow 1mm$ ) as reference solution without adaptivity: mesh-sizes: 2mm; 4mm; 8mm

#### 2D-NURBS elements

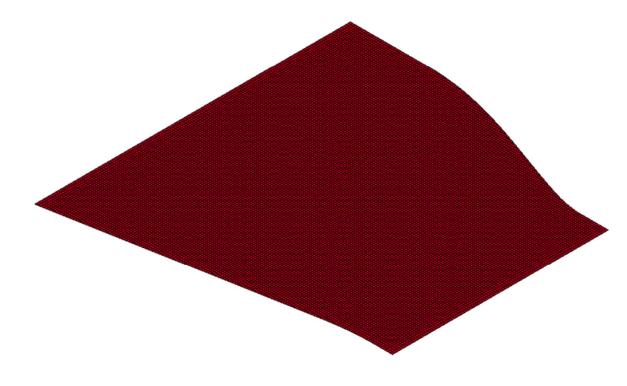
- Formulation: FORM=2 (rotation free formulation)
- Integraion rule: INT=0 (reduced integration)
- Polynomial: p2 (quadratic), p3 (cubic), p4 (quartic), p5 (quintic)
- discretizations: mesh-sizes: 4mm; 8mm; 16mm
- number of interpolation elements/ NURBS-elements: NISR=PR; NISS=PS



#### **Underbody Cross Member** – Standard-Elements

Adaptivity as reference solution

Numisheet2005-BM2 - DL-Adaptiv (4mm -> Time = 0, #nodes=85814, #elem=88980

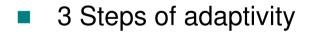


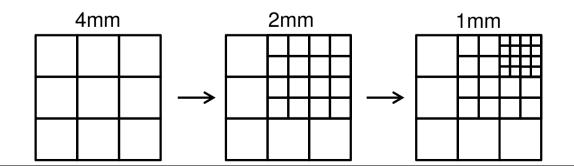




#### **Underbody Cross Member** – Standard-Elements

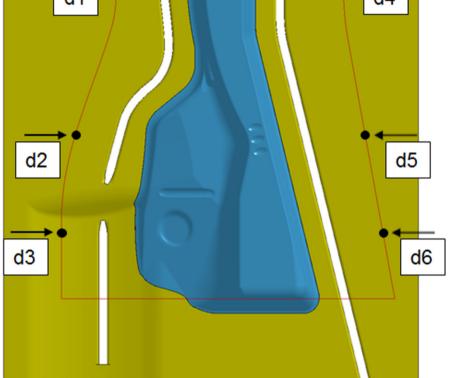
Adaptivity as reference solution

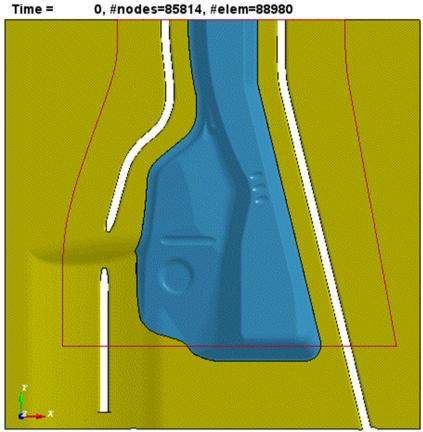






# Underbody Cross Member – Draw-in



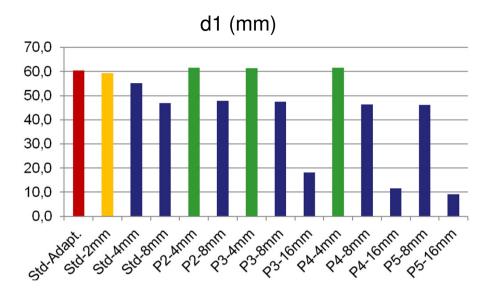


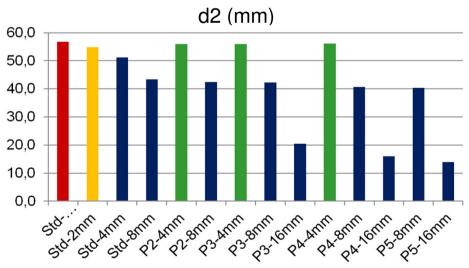
Results  $\rightarrow$  Use "Adaptiv" as reference solution

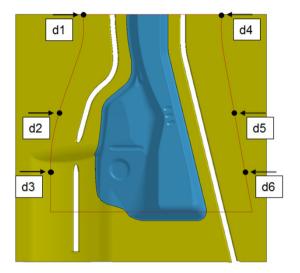
	d1(mm)	d2(mm)	d3(mm)	d4(mm)	d5(5mm)	d6(6mm)
Benchmark	62.2	51.8	56.0	73.7	57.6	47.8
Adaptiv	60.5	56.8	59.9	74.8	57.5	50.4

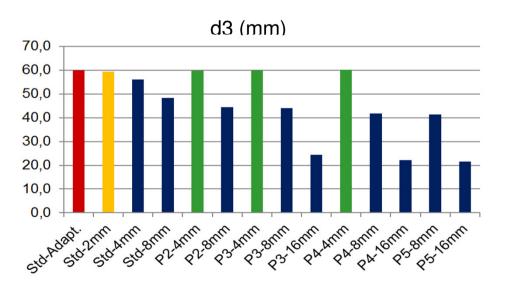


#### Underbody Cross Member – Draw-in





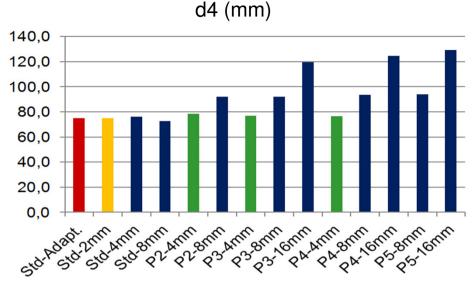




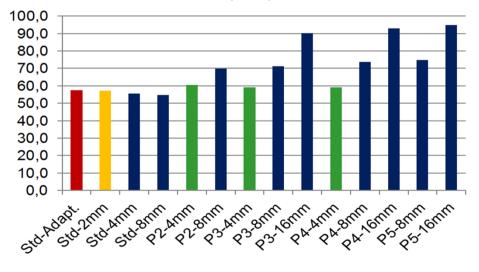
Iarger mesh size → less draw-in (behavior is too stiff)

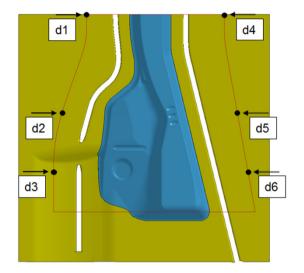


#### Underbody Cross Member – Draw-in

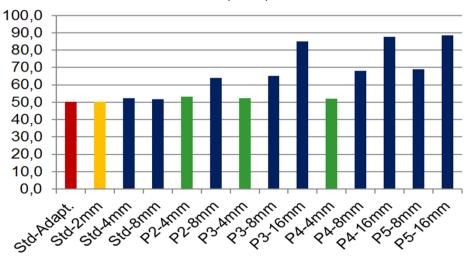




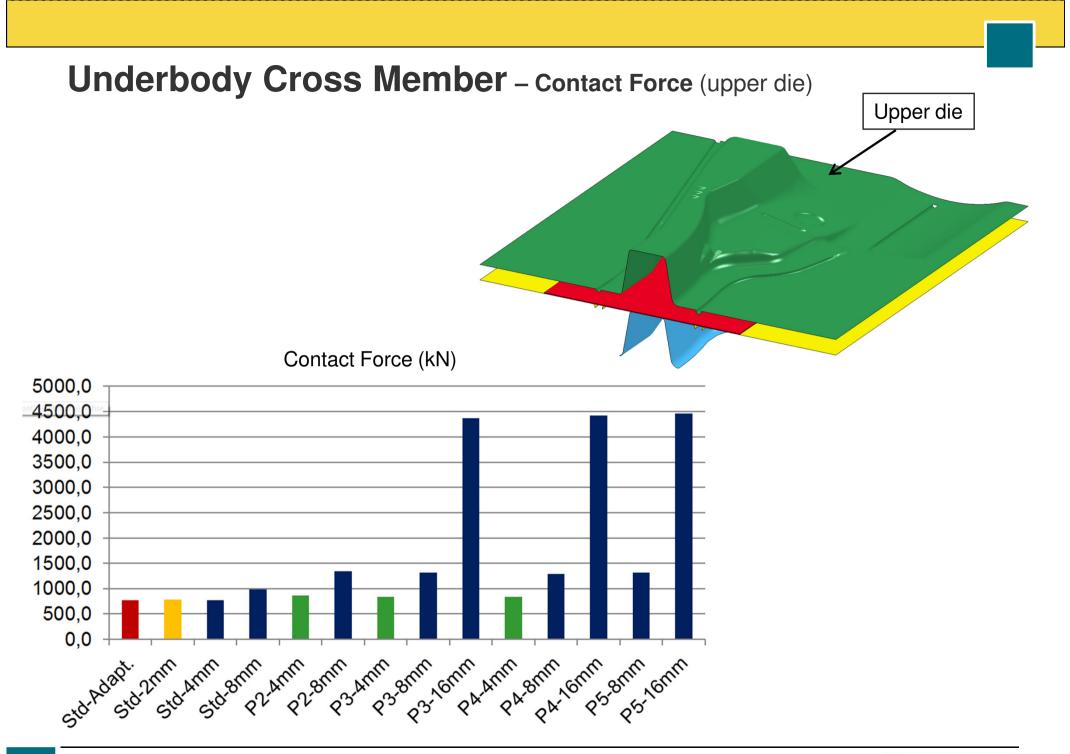




d6 (mm)

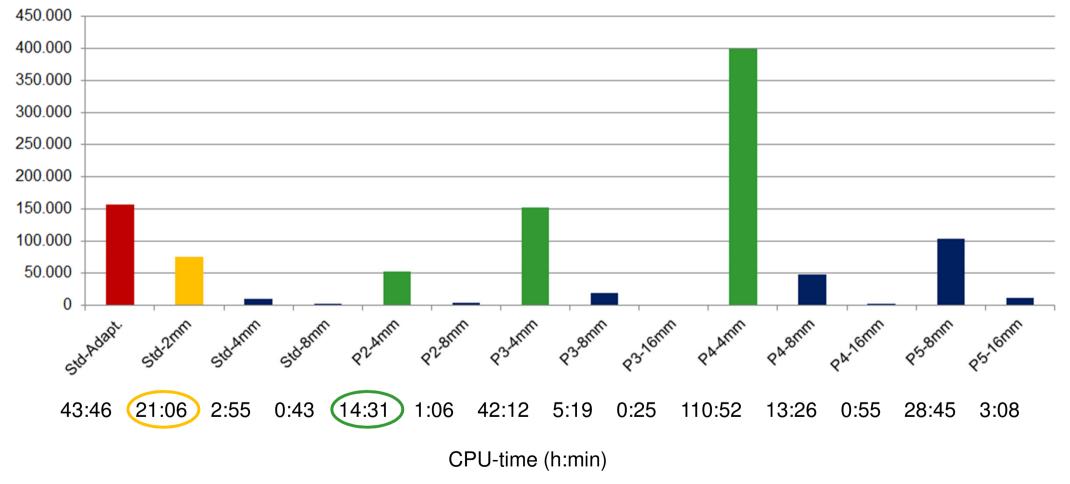






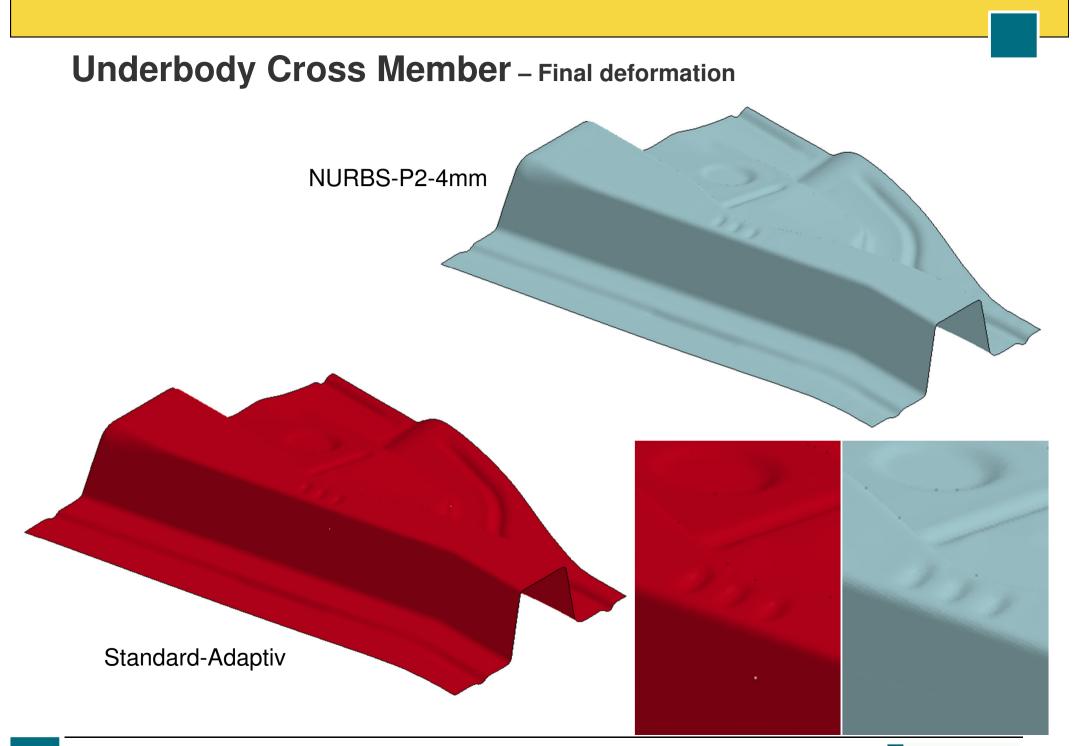


#### **Underbody Cross Member** – CPU-time



CPU-time (s)





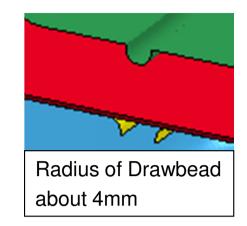
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## Underbody Cross Member – Summary

- Detailed discretization of "Drawbead" needs a fine discretization (<=4mm), no matter what type of elements</li>
- Rotation free elements with reduced integration show best behavior



- CPU-time for comparable discretizations (i.e: p1\_2mm ← → p2\_4mm) are promising (no CODE optimization yet!) → cost competitive
- CPU-time increase for NURBS with same discretizations for next order of polynomial (i.e.: p2\_4mm→p3\_4mm): Factor 2.5-2.8
- Higher order does not help anything in this example (spacing of control points define mesh size)



# Summary

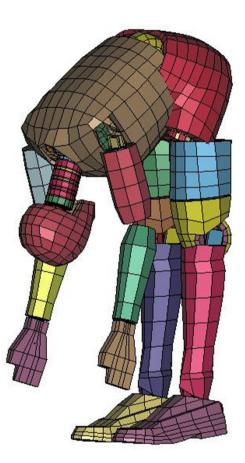
- NURBS-based elements run stable
- higher order accurate isogeometric analysis can be cost competitive
  but missing a couple of "special" issues for industrial sheet metal forming applications
- code optimization necessary to make it faster
- in this example: geometry dictates the mesh size (independent of polynomial order!)

# Outlook

- perform a lot more studies in different fields  $\rightarrow$  experience
- motivate customers (and researchers) to "play" with these elements
- further implementation
  - make NURBS elements work with MPP
  - (selective) mass scaling
  - thickness update of shells
  - use NURBS for contact (instead of interpolation elements)
  - make pre- and post-processing more user-friendly
  - introduce 3D NURBS elements
  - ... much more



# Thank you!



Stefan Hartmann, April 14th, 2011, Pavia, Italy

